

# **MANDATORY BULLETIN**

MB No:

L23/060a REVISION 1

Concerns:

All L 23 SUPER-BLANÍK.

Subject:

Inspection of cable attachment screws and rudder hinge bolt

Reason:

Appearance of crack on the cable attachment screw

To be carried out at the latest by:

Immediately after the bulletin receiving

To be carried out by:

Operator and Organization approved for penetrate test

according to part ASTM E1417-16 (see part A.4)

Costs to be covered by:

Operator

Necessary material to be delivered by:

No material required

Bulletin becomes effective:

On the date of its issue

Total number of pages:

5

Released:

Oldřich Cimbálek ..

anik Aircraft CZ s.r.o.

Chief of Office of Airworthiness Office of Airworthiness

0 4 -03 - 2020

# THE TECHNICAL CONTENT OF THIS DOCUMENT IS APPROVED UNDER THE AUTHORITY OF DOA REF. EASA.21J.609

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L23/060b	IR	6.2.2020	1/6

Beranových 65, 199 00 Praha 9, Czech Republic

# A. WORKING PROCEDURE

# DISMOUNT (to 79. series)

- 1. Dismount the rudder according to the maintenance manual. (page 402-404, 27-20-00)
- 2. Place the rudder on the supports to allow good access to the cable attachment screws pos.21 (2x) and rudder hinge bolt pos. 22. (see figure 1, page 3, 27-20-00 of maintenance manual)
- 3. Clean and degrease the screws and hinge bolt.
- 4. Provide the penetrant testing on the entire accessible shoulder, thread undercut and thread of the screws and hinge bolt according to specification ASTM E1417-16, TYPE II, METHOD C, FORM e. The penetrant test must be provided by an approved organization.
- 5. Acceptable criteria: No cracks in checked area.
- 6. In case of passing the test provide a record in the Airframe Logbook according to the part G of this bulletin. The operation of sailplane is permitted.
- 7. If cracks have been detected on the checked area, the operation of sailplane is prohibited. Send to the sailplane producer:
  - Registration mark
  - Serial Number
  - Total flight hours
  - Number of starts by winch and aerotow

Producer of sailplane will send the bulletin for perform a replacement of the damaged screws.

8. Install the rudder according to the maintenance manual (page 402-404, 27-20-00)

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# DISMOUNT (from 80. series)

- Dismount the rudder according to the maintenance manual. (page 402-404, 27-20-00)
- 2. Place the rudder on the supports to allow good access to the cable attachment screws pos.21 (2x) and rudder hinge bolt pos. 22. (see figure 1, page 3, 27-20-00 of maintenance manual)
- 3. Clean and degrease the stud bolts and hinge bolt. (Do not drill-out the stud bolts out of the rudder).
- 4. Provide the penetrant testing on the entire accessible shoulder, thread undercut and thread of the screws and hinge bolt according to specification ASTM E1417-16, TYPE II, METHOD C, FORM e. The penetrant test must be provided by an approved organization.
- 5. Acceptable criteria: No cracks in checked area.
- 6. In case of passing the test provide a record in the Airframe Logbook according to the part G of this bulletin. The operation of sailplane is permitted.
- 7. If cracks have been detected on the checked area, the operation of sailplane is prohibited. Send to the sailplane producer:
  - Registration mark
  - Serial Number
  - Total flight hours
  - Number of starts by winch and aerotow

Producer of sailplane will send the bulletin for perform a replacement of the damaged screws.

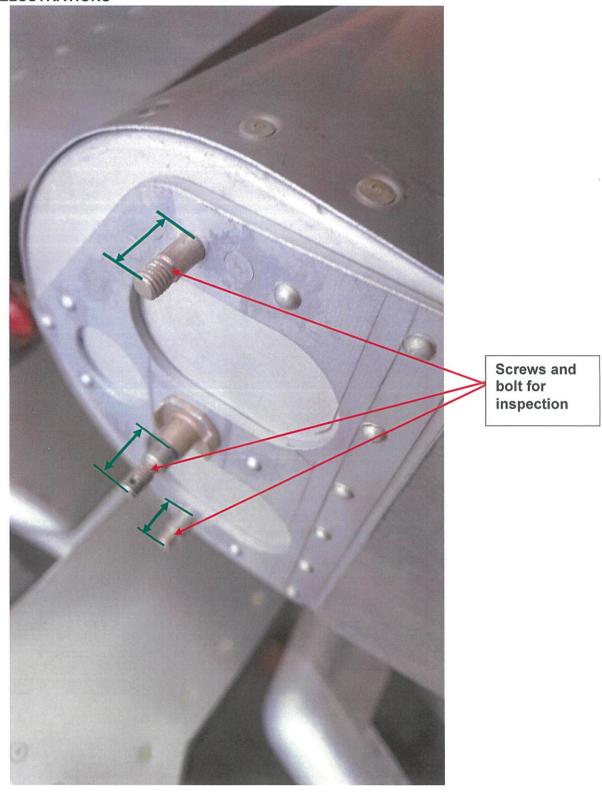
- 8. Install the rudder according to the maintenance manual (page 402-404, 27-20-00)
- 9.

#### **B. MATERIAL REQUIRED**

Not required

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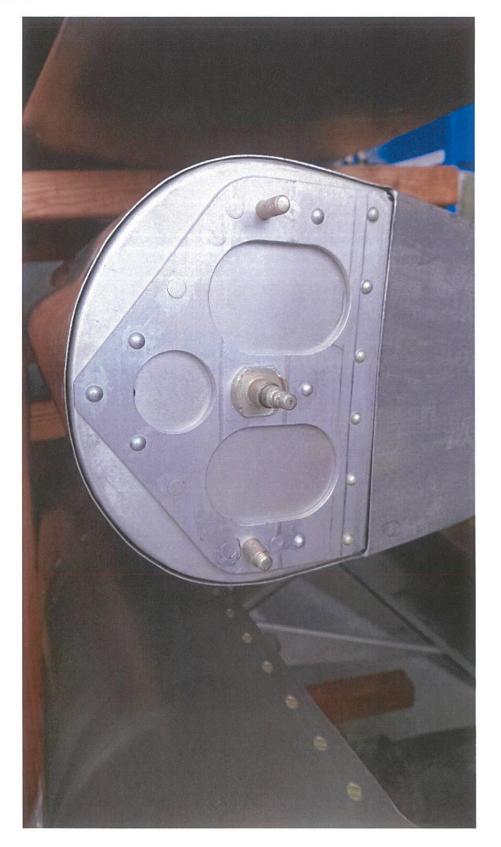
# C. ILLUSTRATIONS



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# Design organization Blanik Aircraft CZ s.r.o. Beranových 65, 199 00 Praha 9, Czech Republic



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# D. DOCUMENTATION REQUIRED

Valid L 23 SUPER-BLANÍK Maintenance manual.

# E. SPARE PARTS IN OPERATION

Not affected

# F. SAILPLANE MASS AND CG

Not affected

#### G. RECORD IN AIRFRAME LOGBOOK AFTER BULLETIN IMPLEMENTATION

Inspection of cable attachment screw and rudder hinge bolt has been performed according to the ZB L23/060a REVISION 1.

- The screws and bolt without defects. The sailplane is released into operation.
- Cracks foung in the (screw/bolt). The sailplane is grounded until the manufacturer has received instructions.

Date:	Carried out by:	
	(legible signature of verification person	n)

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